

Exam 2 Solutions (AST 231 2008 Blackman)

①

$$ds^2 = -\left(1 - \frac{2M}{r}\right) dt^2 + \left(1 - \frac{2M}{r}\right)^{-1} dr^2 + r^2 \sin^2 \theta d\phi^2 + r^2 d\theta^2$$

Q1) at closest approach $r = R$, the turning point, there $\left. \frac{dr}{d\tau} \right|_R = 0$. From 9.21, 9.22:

Ⓐ $\frac{d\phi}{d\tau} = \frac{L}{r^2}$ (for $\sin\theta = \frac{\pi}{2}$)

• locally, SR holds so energy measured by observer is

$$E = \frac{m_a}{(1-v^2)^{1/2}} = -p \cdot U_{\text{obs}} \quad (1)$$

4-velocity of asteroid

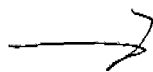
Energy measured by observer with 4-velocity U_{obs}

• for stationary observer at R

$$g_{\mu\nu} U_{\text{obs}}^\mu U_{\text{obs}}^\nu = -1 = g_{00} U_{\text{obs}}^{t^2} = \left(1 - \frac{2M}{R}\right) U_{\text{obs}}^{0^2}$$

$$\Rightarrow U_{\text{obs}}^t = \left(1 - \frac{2M}{R}\right)^{-1/2}$$

$$U_{\text{obs}}^\mu = \left(\left(1 - \frac{2M}{R}\right)^{-1/2}, 0, 0, 0 \right)$$



$$\Rightarrow E = -g_{tt} p^t u_{obs}^t = p^t \left(1 - \frac{2M}{R}\right)^{1/2} \quad (2)$$

$$= \underbrace{m_a}_{\text{asteroid mass}} u^t \left(1 - \frac{2M}{R}\right)^{1/2} \quad (2)$$

Since $u \cdot u = -1 = g_{\mu\nu} u^\mu u^\nu$

We can obtain u^t : (at $r=R$, have only u^t & u^ϕ)
use $\sin\theta = \frac{b}{r}$

$$-\left(1 - \frac{2M}{R}\right) u^t{}^2 + R^2 u^\phi{}^2 = -1$$

but $u^\phi = \frac{d\phi}{d\tau} = \frac{l}{r^2}$ from (A) so at $r=R$

$$\Rightarrow -\left(1 - \frac{2M}{R}\right) u^t{}^2 + R^2 \frac{l^2}{R^4} = -1$$

$$\Rightarrow u^t = \frac{\left(1 + \frac{l^2}{R^2}\right)^{1/2}}{\left(1 - \frac{2M}{R}\right)^{1/2}} \Rightarrow V = \frac{l/R}{\left[1 + \left(\frac{l}{R}\right)^2\right]^{1/2}} \quad (3)$$

• need to get $l(b, R)$

• at large $r \gg R$, impact parameter $b = r\phi$

$$\text{so there: } l \equiv r^2 \frac{d\phi}{d\tau} = -r^2 \frac{d\phi}{dr} \frac{dr}{d\tau} = r^2 \left(\frac{b}{r^2}\right) \frac{dr}{d\tau} = -b \frac{dr}{d\tau} \quad (4)$$

But at large r , V_{eff} in eq 9.29 vanishes

$$\text{so } \mathcal{E} = \frac{e^2 - 1}{2} = \left(\frac{dr}{d\tau}\right)^2 / 2 \Rightarrow \frac{dr}{d\tau} = (e^2 - 1)^{1/2} \text{ at large } r.$$

This relates l to e in (4): \Rightarrow

$$l = +b(e^2 - 1)^{1/2} \quad (5)$$

\uparrow since $\frac{dr}{d\tau} < 0$.

now turning point condition from

equation 9.29: $\frac{dr}{dt} = 0$ at $r=R$ so from 9.28:

(3)

$$\epsilon \equiv \frac{e^2 - 1}{2} = V_{\text{eff}}(R) = \frac{1}{2} \left[\left(1 - \frac{2M}{R}\right) \left(1 + \frac{e^2}{R^2}\right) - 1 \right]$$

$$e^2 - 1 = \left(1 - \frac{2M}{R}\right) \left(1 + \frac{e^2}{R^2}\right) - 1 \quad \text{at turning point}$$

$$\begin{aligned} \Rightarrow \frac{e^2}{R^2} &= \frac{e^2}{R^2} - \frac{2M}{R} - \frac{2Me^2}{R^3} \\ &= \frac{e^2}{R^2} - \frac{2M}{R} - \frac{2Me^2}{R^3} \end{aligned} \quad (6)$$

Solving (6) & (5) for e^2/R^2

$$\Rightarrow \frac{e^2}{R^2} = \frac{b^2}{R^2} \left(\frac{e^2}{R^2} - \frac{2M}{R} - \frac{2Me^2}{R^3} \right) = \frac{e^2}{R^2} \left(\frac{b^2}{R^2} - \frac{2Mb^2}{R^3} \right) - \frac{2Mb^2}{R^3}$$

$$\Rightarrow \frac{e^2}{R^2} = \frac{2Mb^2/R^3}{\frac{b^2}{R^2} - \frac{2Mb^2}{R^3} - 1} \quad \text{now plug into (3):}$$

$$\Rightarrow V^2 = \frac{1}{(R/e)^2 + 1} = \frac{2Mb^2/R^3}{\frac{b^2}{R^2} - \frac{2Mb^2}{R^3} - 1 + 2Mb^2/R^3}$$

$$\Rightarrow V = \frac{\left(\frac{2M}{R}\right)^{1/2} \frac{b}{R}}{\left(\frac{b^2}{R^2} - 1\right)^{1/2}}$$

Q2) $U \cdot U = g_{\alpha\beta} U^\alpha U^\beta$

$$\begin{aligned} \frac{d(U \cdot U)}{d\tau} &= g_{\alpha\beta} U^\alpha \frac{dU^\beta}{d\tau} + g_{\alpha\beta} \frac{dU^\alpha}{d\tau} U^\beta \\ &+ \frac{dg_{\alpha\beta}}{d\tau} U^\alpha U^\beta \quad \text{via chain rule} \\ &= 2g_{\alpha\beta} U^\alpha \frac{dU^\beta}{d\tau} + \frac{\partial g_{\alpha\beta}}{\partial x^\mu} U^\mu U^\alpha U^\beta \\ &= 2g_{\alpha\beta} U^\alpha \frac{dU^\beta}{d\tau} + \frac{\partial g_{\alpha\beta}}{\partial x^\mu} U^\mu U^\alpha U^\beta \end{aligned}$$

geodesic equation 8.15: $\frac{dU^\beta}{d\tau} = -\Gamma_{\alpha\sigma}^\beta U^\alpha U^\sigma$

\Rightarrow

$$\begin{aligned} &= -2g_{\alpha\beta} U^\alpha \Gamma_{\sigma\alpha}^\beta U^\sigma U^\alpha + \frac{\partial g_{\alpha\beta}}{\partial x^\mu} U^\mu U^\alpha U^\beta \\ &= U^\alpha U^\sigma \left(-2g_{\alpha\beta} U^\alpha \Gamma_{\sigma\alpha}^\beta + \frac{\partial g_{\alpha\beta}}{\partial x^\mu} U^\mu \right) \end{aligned}$$

from 8.14: $g_{\alpha\beta} \Gamma_{\sigma\alpha}^\beta = \frac{1}{2} \left(\frac{\partial g_{\alpha\gamma}}{\partial x^\sigma} + \frac{\partial g_{\alpha\sigma}}{\partial x^\gamma} - \frac{\partial g_{\sigma\gamma}}{\partial x^\alpha} \right)$

$$\begin{aligned} \Rightarrow & U^\alpha U^\sigma \left(-U^\alpha \frac{\partial g_{\alpha\gamma}}{\partial x^\sigma} + U^\alpha \frac{\partial g_{\alpha\sigma}}{\partial x^\gamma} - U^\alpha \frac{\partial g_{\sigma\gamma}}{\partial x^\alpha} + \frac{\partial g_{\sigma\gamma}}{\partial x^\mu} U^\mu \right) \\ &= -U^\alpha U^\sigma U^\alpha \frac{\partial g_{\alpha\gamma}}{\partial x^\sigma} + U^\alpha U^\sigma U^\alpha \frac{\partial g_{\alpha\sigma}}{\partial x^\gamma} \quad \text{cancel} \\ &= \boxed{0} \end{aligned}$$

(remember: repeated indices sum & $g_{\alpha\beta}$ is symmetric)

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Q3) $-dt^2 + a^2(t)(dx^2 + dy^2 + dz^2)$

metric coefficients are independent of x, y, z . 3-killing vectors corresponding to each of these directions.

Consider propagation in x direction; associated killing vector is $\sum_x^m (0, 1, 0, 0)$.

Let \underline{p} be photon 4-momentum.

Then $\sum_x \cdot \underline{p}$ is conserved. \therefore so

$$\sum_x \cdot \underline{p} = g_{\mu\nu} \sum_x^m p^\nu = g_{xx} p^x(t) = a^2(t) p^x(t) = \text{constant}$$

* since \underline{p} is null: $\Rightarrow p^x \propto \frac{1}{a^2(t)} \quad (1)$

$$\underline{p} \cdot \underline{p} = 0 = -p^{t^2} + a^2(t) p^{x^2} = 0$$

$$\Rightarrow p^t(t) = a(t) p^x \Rightarrow p^t \propto \frac{1}{a(t)} \quad \text{(from (1))} \quad (2)$$

\therefore stationary observer has $u_{\text{obs}} = (1, 0, 0, 0)$ in this metric and measures energy

$$E = -\underline{p} \cdot u_{\text{obs}} = -g_{tt} p^t = +p^t(t) \propto \frac{1}{a(t)} \quad \text{from (2).}$$

$$\Rightarrow \frac{E(t)}{E(t_0)} = \frac{w(t)}{w(t_0)} = \frac{a(t_0)}{a(t)} = \text{cosmological redshift}$$

(6)

(Q4) from box 11-1 d or e

$$4\pi R^2 \sigma T^4 = \epsilon L_{\text{edd}}$$

$$\Rightarrow T = 5 \times 10^7 \left(\frac{GM}{c^2 R} \right)^{1/2} \left(\epsilon \frac{M_0}{M} \right)^{1/4} \text{ K}$$

$$\omega = 6 \times 10^{10} \text{ T (Hz)}$$

$$\Rightarrow \omega = 3 \times 10^{18} \left(\frac{GM}{c^2 R} \right)^{1/2} \left(\epsilon \frac{M_0}{M} \right)^{1/4} \text{ Hz}$$

for $M = 10^9 M_0$, $R = R_{\text{ISCO}} = 6 \frac{GM}{c^2}$, $\epsilon = 0.1$

$$\Rightarrow \omega = 3 \times 10^{18} \left(\frac{1}{6} \right)^{1/2} (0.1)^{1/4} (10^{-9})^{1/4} \text{ Hz}$$

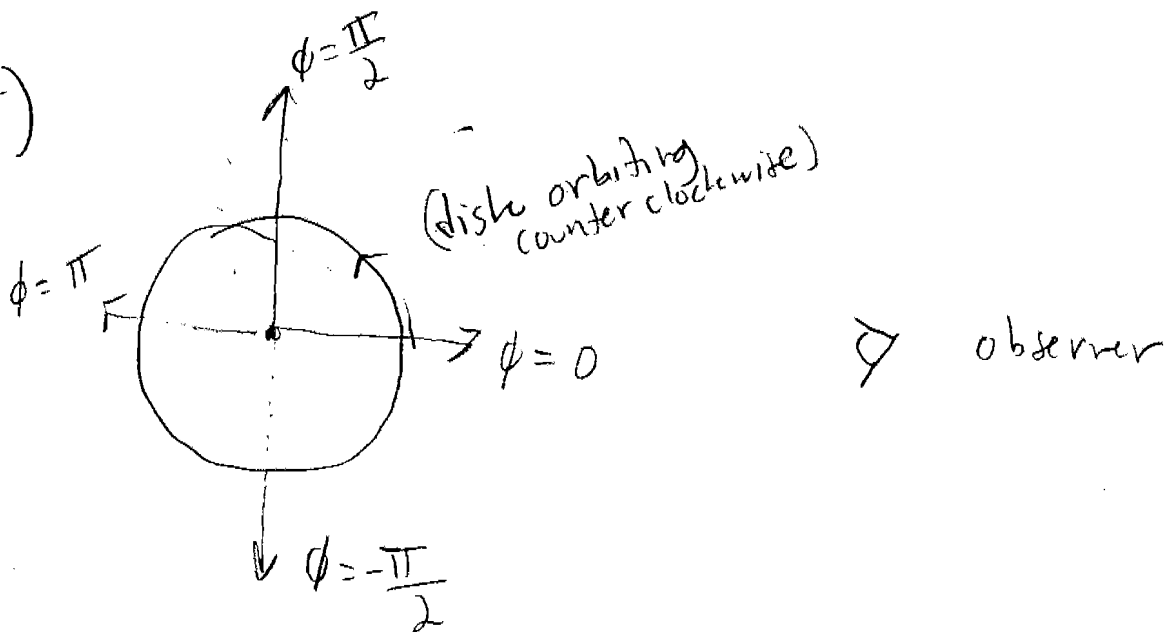
$$= \underline{3 \times 10^{15} \text{ Hz}} \quad (\text{ultraviolet})$$

• for $M = 10 M_0$, $\epsilon = 0.1$, $R = R_{\text{ISCO}} = 6 \frac{GM}{c^2}$:

$$\Rightarrow \omega = (3 \times 10^{15}) \left(\frac{10^9}{10} \right)^{1/4} = \underline{3 \times 10^{17} \text{ Hz}}$$

(x-rays)

(Q5)



(7)

• at $\phi = 0$ photons have no motion along line of sight so there is only gravitational redshift. At $\phi = \frac{\pi}{2}$, disk is moving AWAY from observer so the gravitational redshift and doppler shift both act to reduce the frequency. $\phi = \frac{\pi}{2}$ corresponds to minimum.

At $\phi = -\frac{\pi}{2}$ doppler shift increases frequency competing with the effect of grav. redshift. $\phi = -\frac{\pi}{2}$ would thus produce largest possible frequency observed at ∞



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From Eqn. 11-24 for $|\phi| = \frac{\pi}{2}$

$$\frac{\omega_o}{\omega_{em}} = \frac{\left(1 - \frac{3M}{r}\right)^{1/2}}{\left(1 \pm \frac{1}{\left(\frac{r}{m} - 2\right)^{1/2}}\right)}$$

the minus sign must correspond to $\phi = -\frac{\pi}{2}$ since it gives the largest value.

at $r = 6M$

$$\frac{\omega_o}{\omega_{em}} = \frac{\left(1 - \frac{1}{2}\right)^{1/2}}{\left(1 - \frac{1}{2}\right)} = \frac{2}{\sqrt{2}} = \boxed{\sqrt{2}}$$

b) since $\sqrt{2} > 1 \Rightarrow$ Doppler effect dominates the gravitational redshift effect; Doppler effect of $\phi = -\frac{\pi}{2}$ acts to increase observed frequency from that emitted while grav. redshift acts to lower frequency. $\frac{\omega_o}{\omega_{em}} > 1$ means Doppler wins.